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LIFE-CRITICAL DIGITAL FLIGHT CONTROL SYSTEMS

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N91-17561

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LIFE-CRITICAL DIGITAL FLIGHT CONTROL SYSTEMS

INDUSTRY STATUS

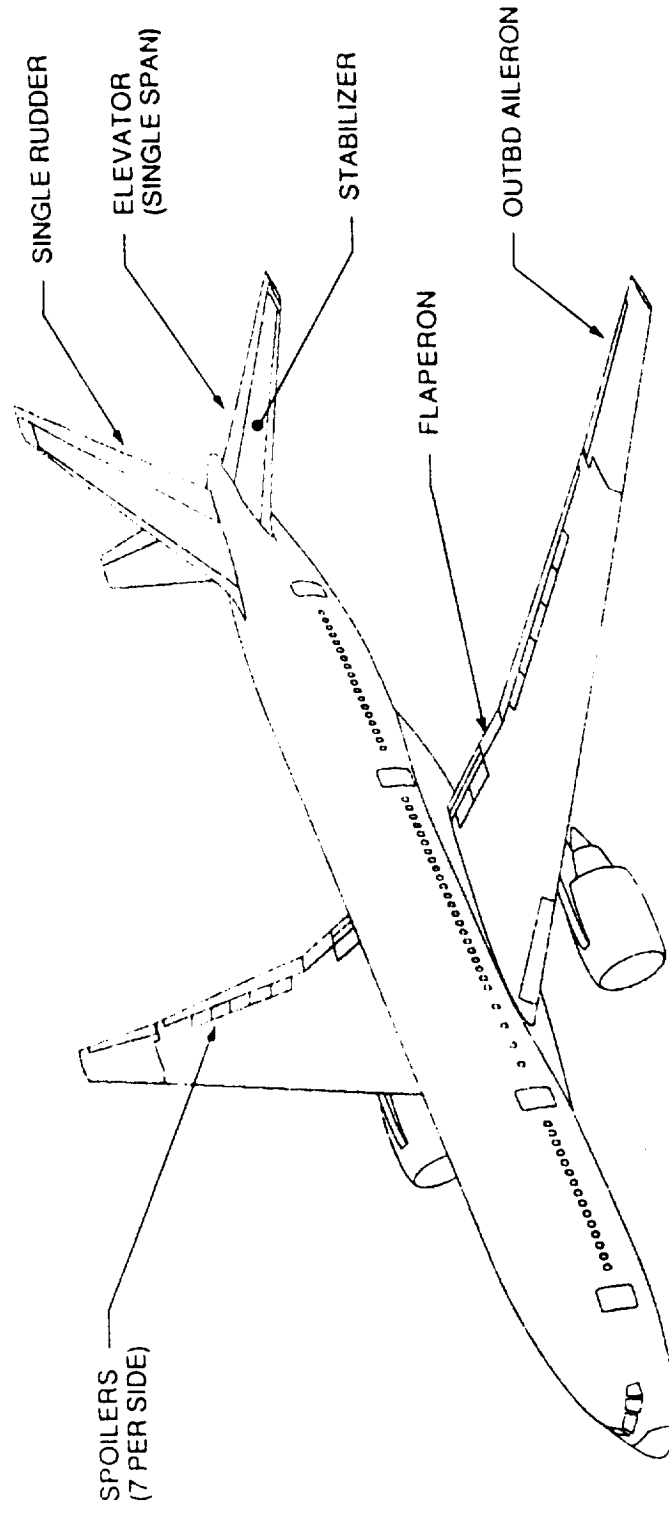
DIGITAL AUTOPILOT SYSTEMS WERE FIRST CERTIFICATED FOR USE ON COMMERCIAL AIRPLANES IN THE LATE 1970'S

THE A-320 AIRPLANE WAS THE FIRST COMMERCIAL AIR TRANSPORT AIRPLANE TO BE CERTIFICATED WITH A FLY BY WIRE PRIMARY FLIGHT CONTROL SYSTEM

BOEING WILL HAVE ALL FLY BY WIRE FLIGHT CONTROLS ON THE 767-X (777) AIRPLANE

LIFE-CRITICAL DIGITAL FLIGHT CONTROL SYSTEMS

- DEFINITION**
- SAFETY**
- INDUSTRY STATUS**
- PROGRAM PHASES**



767-X PRIMARY FLIGHT CONTROL SURFACES

LIFE-CRITICAL DIGITAL FLIGHT CONTROL SYSTEMS

DEFINITION

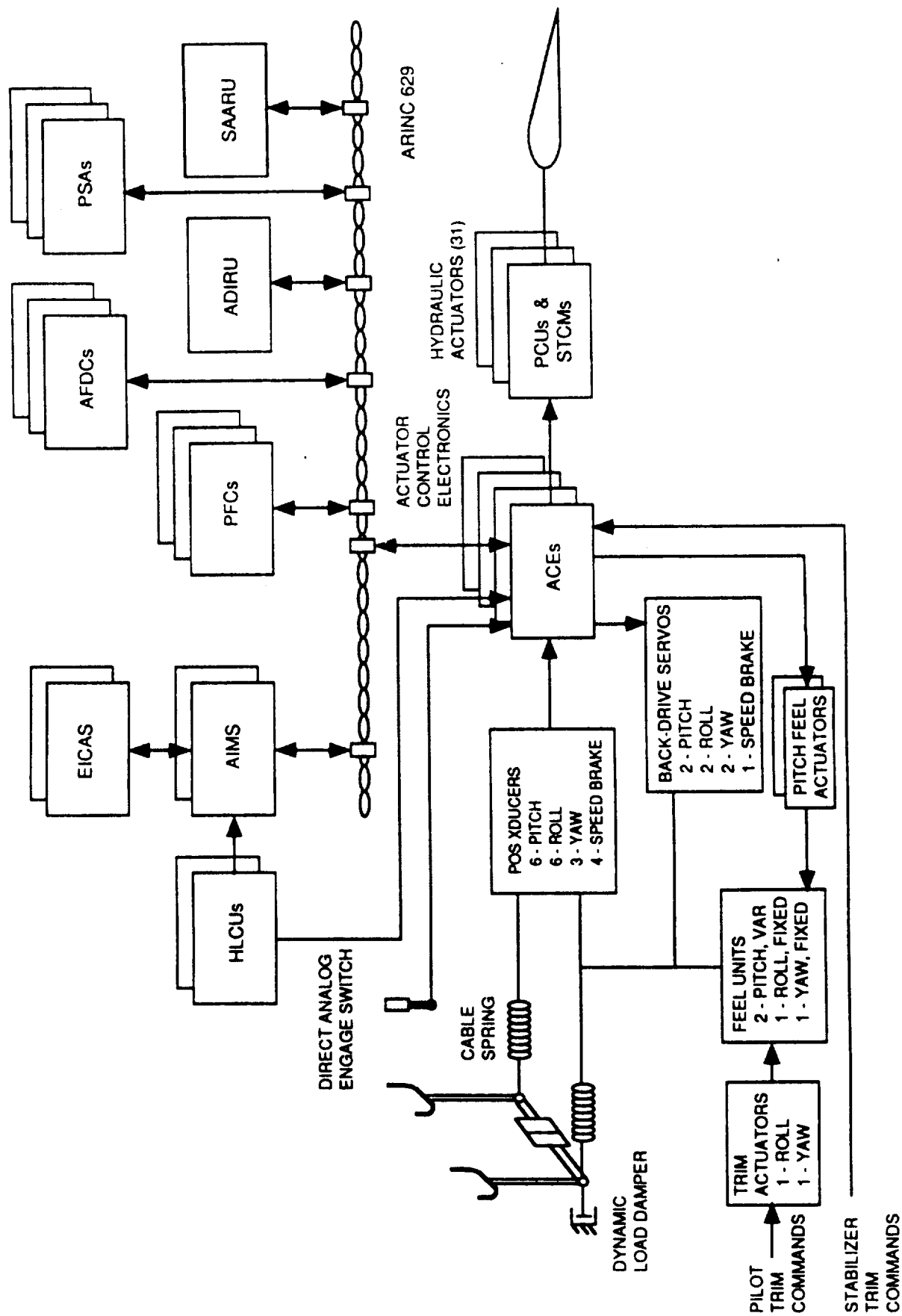
A CONTROL SYSTEM IMPLEMENTED IN DIGITAL COMPUTER TECHNOLOGY WHICH HAS A FUNCTION WHICH IF NOT PERFORMED AS INTENDED IS LIFE THREATENING

EXAMPLES: **AN AUTOPILOT USED FOR AUTOMATIC LANDING IN LOW
VISIBILITY CONDITIONS**

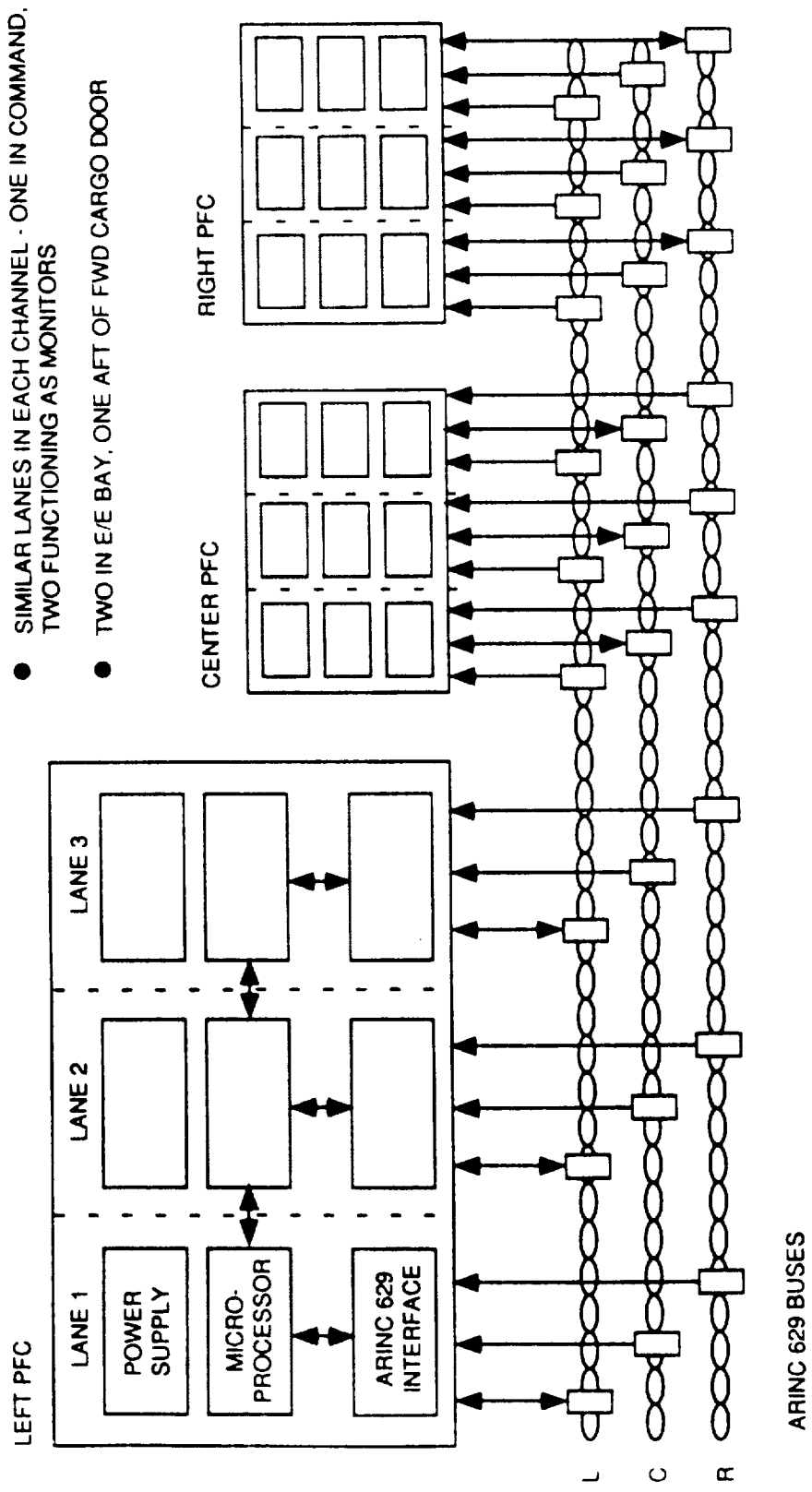
**AN AIRPLANE CONTROL SYSTEM IMPLEMENTED WITHOUT
CONTROL CABLES:**

FLY BY WIRE

FLY BY LIGHT



767-X PRIMARY FLIGHT CONTROL SYSTEM OVERVIEW

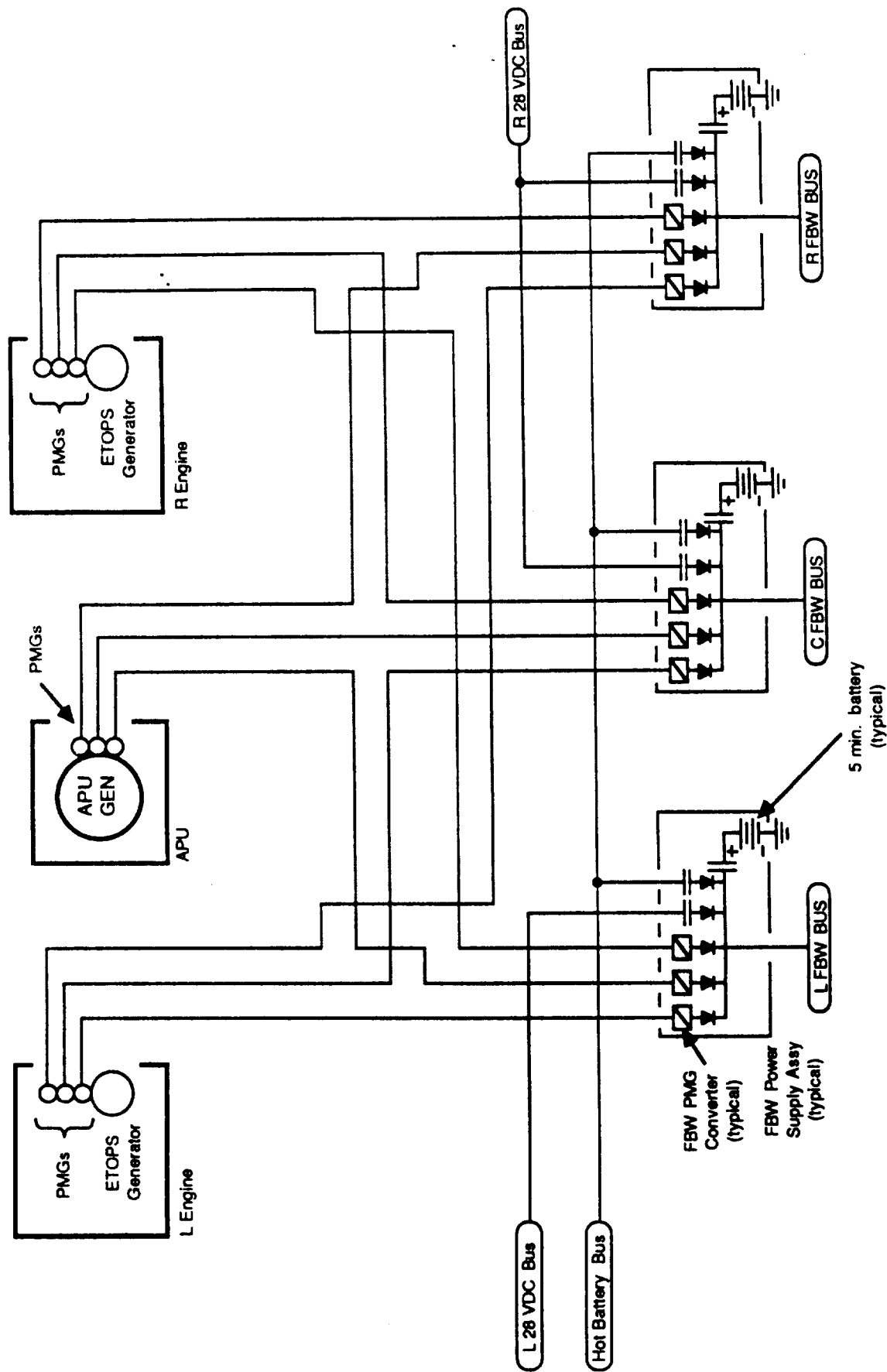


- DISSIMILAR CHANNELS - LEFT, CENTER, RIGHT
- SIMILAR LANES IN EACH CHANNEL - ONE IN COMMAND, TWO FUNCTIONING AS MONITORS
- TWO IN E/E BAY, ONE AFT OF FWD CARGO DOOR

PRIMARY FLIGHT COMPUTER ARCHITECTURE

767-X ELECTRICAL POWER SYSTEM

FLIGHT CRITICAL DC





767-X PRIMARY FLIGHT CONTROLS HYDRAULIC / ACE DISTRIBUTION

LIFE-CRITICAL DIGITAL FLIGHT CONTROL SYSTEMS

SAFETY

FEDERAL AVIATION ADMINISTRATION (FAA) REGULATIONS DEFINE THE BASIC SAFETY CRITERIA:

FAR 25.1309 NO SINGLE FAILURE OR COMBINATION OF FAILURES WHICH ARE NOT SHOWN TO BE EXTREMELY IMPROBABLE SHALL PREVENT CONTINUED SAFE FLIGHT AND LANDING OF THE AIRPLANE

EXTREMELY IMPROBABLE - PROBABILITY OF 1×10^{-9} OR LESS PER FLIGHT HOUR OR
EVENT

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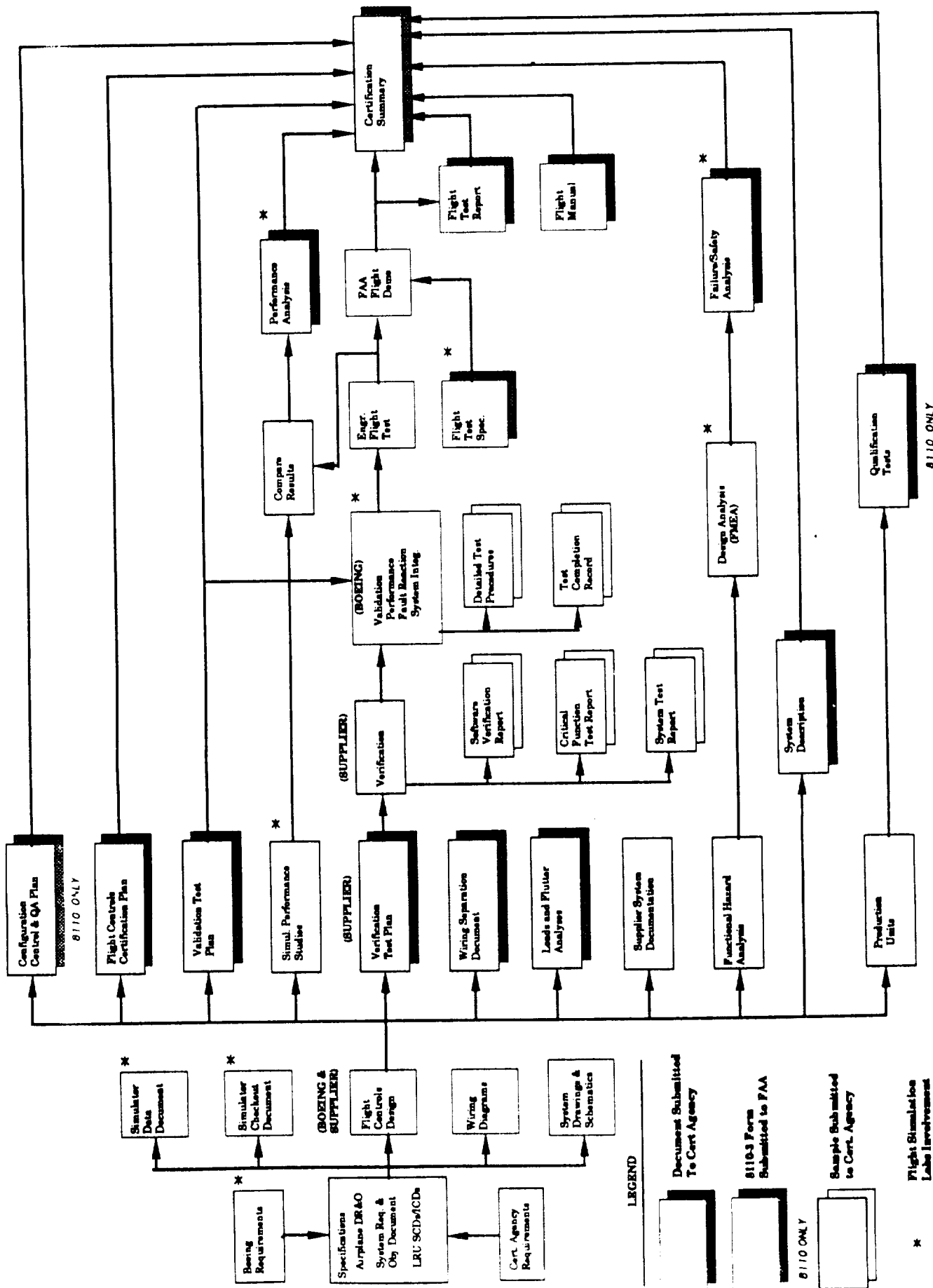


FIGURE 1.2-1 PRIMARY FLIGHT CONTROLS DEVELOPMENT OVERVIEW

LIFE-CRITICAL DIGITAL FLIGHT CONTROL SYSTEMS

PROGRAM PHASES - REQUIREMENTS DEFINITION

TOP DOWN STRUCTURED PROCESS:

AIRPLANE LEVEL REQUIREMENTS	TOP LEVEL DESIGN REQUIREMENTS AND OBJECTIVES
<p>1. The airplane shall be capable of operating from runways with a length of 1,500 feet or less.</p> <p>2. The airplane shall be capable of operating from runways with a width of 60 feet or less.</p> <p>3. The airplane shall be capable of operating from runways with a surface strength of 15,000 psi or less.</p> <p>4. The airplane shall be capable of operating from runways with a surface temperature of 120 degrees Fahrenheit or less.</p> <p>5. The airplane shall be capable of operating from runways with a surface condition of dry, wet, or icy.</p> <p>6. The airplane shall be capable of operating from runways with a surface condition of smooth, rough, or uneven.</p> <p>7. The airplane shall be capable of operating from runways with a surface condition of hard, soft, or sandy.</p> <p>8. The airplane shall be capable of operating from runways with a surface condition of level, sloped, or curved.</p> <p>9. The airplane shall be capable of operating from runways with a surface condition of straight, curved, or irregular.</p> <p>10. The airplane shall be capable of operating from runways with a surface condition of flat, hilly, or mountainous.</p>	<p>1. The airplane shall be capable of operating from runways with a length of 1,500 feet or less.</p> <p>2. The airplane shall be capable of operating from runways with a width of 60 feet or less.</p> <p>3. The airplane shall be capable of operating from runways with a surface strength of 15,000 psi or less.</p> <p>4. The airplane shall be capable of operating from runways with a surface temperature of 120 degrees Fahrenheit or less.</p> <p>5. The airplane shall be capable of operating from runways with a surface condition of dry, wet, or icy.</p> <p>6. The airplane shall be capable of operating from runways with a surface condition of smooth, rough, or uneven.</p> <p>7. The airplane shall be capable of operating from runways with a surface condition of hard, soft, or sandy.</p> <p>8. The airplane shall be capable of operating from runways with a surface condition of level, sloped, or curved.</p> <p>9. The airplane shall be capable of operating from runways with a surface condition of straight, curved, or irregular.</p> <p>10. The airplane shall be capable of operating from runways with a surface condition of flat, hilly, or mountainous.</p>

SYSTEM REQUIREMENTS

CERTIFICATION REQUIREMENTS

FUNCTIONAL REQUIREMENTS

INTEGRITY REQUIREMENTS

ARCHITECTURAL CONSIDERATIONS

SOFTWARE REQUIREMENTS

LIFE-CRITICAL DIGITAL FLIGHT CONTROL SYSTEMS

PROGRAM PHASES - REQUIREMENTS DEFINITION

TOP DOWN STRUCTURED PROCESS:

AIRPLANE LEVEL REQUIREMENTS TOP LEVEL DESIGN REQUIREMENTS AND OBJECTIVES

SYSTEM REQUIREMENTS CERTIFICATION REQUIREMENTS
FUNCTIONAL REQUIREMENTS
INTEGRITY REQUIREMENTS
ARCHITECTURAL CONSIDERATIONS

SOFTWARE REQUIREMENTS EXPANSION OF SYSTEM REQUIREMENTS TO A
LEVEL WHICH CAN BE IMPLEMENTED IN A TARGET
DIGITAL COMPUTER OR COMPUTERS

LIFE-CRITICAL DIGITAL FLIGHT CONTROL SYSTEMS

PROGRAM PHASES - DESIGN AND DEVELOPMENT

HARDWARE SELECTION	I/O REQUIREMENTS PROCESSING SPEED MEMORY SIZE ETC
PROGRAMMING LANGUAGE	INDUSTRY/COMPANY STANDARD SUPPORT SOFTWARE AVAILABILITY AND MATURITY LONG TERM MAINTENANCE ETC
CODE GENERATION	TYPICALLY AN INCREMENTAL BUILD PROCESS
TESTING	HARDWARE - QUALIFICATION TESTING - RTCA DO-160 INCREMENTAL SOFTWARE LOADS - VENDOR AND AIRFRAME SYSTEMS INTEGRATION / IRON BIRD AIRPLANE - GROUND AND FLIGHT

LIFE-CRITICAL DIGITAL FLIGHT CONTROL SYSTEMS

PROGRAM PHASES - VERIFICATION

GUIDELINE DOCUMENT RTCA DOCUMENT DO-178A

VERIFICATION PROCESSES ARE A FUNCTION OF SYSTEM CRITICALITY

CRITICAL SYSTEM	A FORMAL PROCESS OF ASSURING THAT ALL SOFTWARE REQUIREMENTS HAVE BEEN IMPLEMENTED <u>COMPLETELY</u> AND <u>EXCLUSIVELY</u>
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LIFE-CRITICAL DIGITAL FLIGHT CONTROL SYSTEMS

PROGRAM PHASES - VALIDATION

**A PROCESS OF ASSURING THAT ALL SYSTEM REQUIREMENTS HAVE BEEN
IMPLEMENTED CORRECTLY**

o ANALYSES

SAFETY ANALYSIS

**HAZARD ASSESSMENT AND FAILURE
ANALYSIS TO ASSURE THAT REQUIREMENTS
OF FAR 25.1309 ARE SATISFIED**

PERFORMANCE ANALYSIS

**ASSURANCE THAT SYSTEM PERFORMS
INTENDED FUNCTION WITHIN ACCEPTABLE
LIMITS UNDER ALL ALLOWABLE
ENVIRONMENTAL AND TOLERANCE
CONDITIONS**

LIFE-CRITICAL DIGITAL FLIGHT CONTROL SYSTEMS

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LIFE-CRITICAL DIGITAL FLIGHT CONTROL SYSTEMS

PROGRAM PHASES - CERTIFICATION

THE PROCESS OF DEMONSTRATING TO THE REGULATORY AUTHORITIES THAT ALL SAFETY AND PERFORMANCE REQUIREMENTS ARE SATISFIED

STARTS WITH A CERTIFICATION PLAN WHICH:

IDENTIFIES REGULATIONS AND ACCEPTABLE MEANS OF COMPLIANCE METHODS

DESCRIBES PROPOSED METHODS OF ESTABLISHING COMPLIANCE

DESCRIBES THE METHODS AND PROCESSES TO BE USED TO ASSURE AN ORDERLY AND CONTROLLED DESIGN AND DEVELOPMENT PROCESS

FOLLOW ON SPECIALIST MEETINGS

PERFORMANCE AND INTEGRITY DEMONSTRATIONS

LIFE-CRITICAL DIGITAL FLIGHT CONTROL SYSTEMS

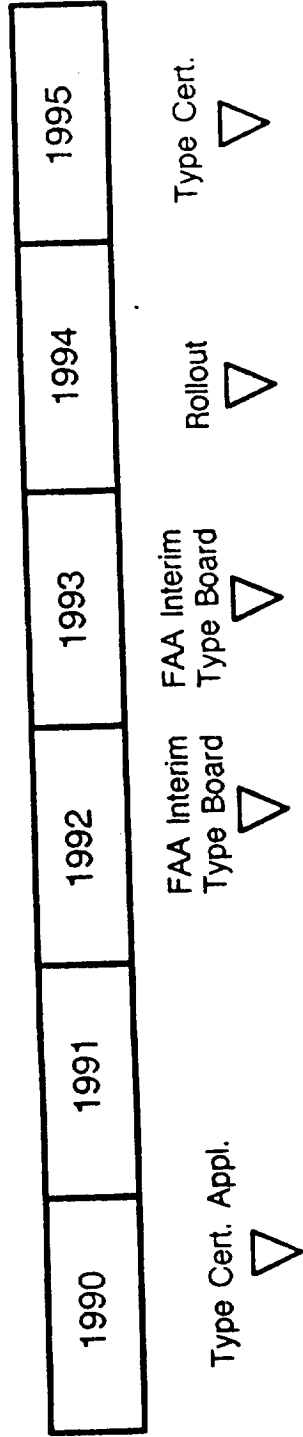
PROGRAM PHASES - CERTIFICATION (CONT)

CERTIFICATION SUMMARY

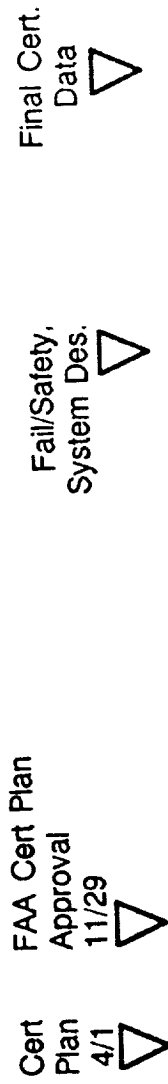
**CONFIRMS COMPLETE IMPLEMENTATION OF THE PROCESSES IDENTIFIED IN THE
CERTIFICATION PLAN**

**PROVIDES A MEANS FOR ESTABLISHING VERIFICATION AND VALIDATION
COVERAGE**

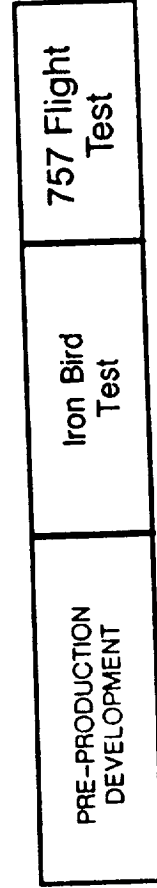
767-X PFCs Schedule



767-X

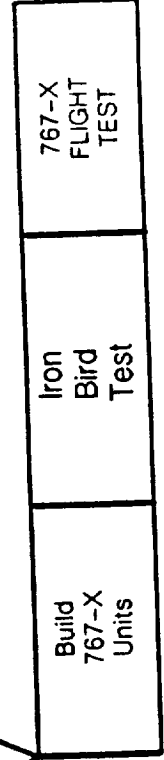


Certification Documentation



757 PFCs Testing

Production Design Validation 



767-X PFCs